



**Efficient Solution of the
Long-Rod Penetration Equations
of Alekseevskii-Tate**

by Steven B. Segletes and William P. Walters

ARL-TR-2855

September 2002

Army Research Laboratory

Aberdeen Proving Ground, MD 21005-5066

ARL-TR-2855

September 2002

Efficient Solution of the Long-Rod Penetration Equations of Alekseevskii-Tate

**Steven B. Segletes and William P. Walters
Weapons and Materials Research Directorate, ARL**

Contents

1. Background	1
2. Closed Form Solution for $L(V)$	2
3. Choice of Model Variable	4
4. Model-Variable Transformation	5
5. Penetration	6
5.1 $R = Y$	6
5.2 $\gamma = 1$	7
5.3 General Case.....	8
6. Implicit Time	12
7. Terminal Rod Length, etc.	13
8. Residual Erosion/Penetration Behaviors	14
8.1 Residual Rod Erosion.....	15
8.2 Residual Rigid-Body Penetration.....	16
9. Conclusions	17
10. References	18
Report Documentation Page	19

1. Background

The penetration equations that describe the behavior of a long rod that erodes while it penetrates at high velocity were formulated independently by Alekseevskii [1] and Tate [2] in the mid-1960s and are given by

$$L\dot{V} = -Y/\rho_R \quad (\text{rod deceleration}), \quad (1)$$

$$\frac{1}{2}\rho_R(V-U)^2 + Y = \frac{1}{2}\rho_T U^2 + R \quad (\text{interface stress balance}), \quad (2)$$

$$V = U - \dot{L} \quad (\text{erosion kinematics}), \quad \text{and} \quad (3)$$

$$\dot{P} = U \quad (\text{penetration definition}). \quad (4)$$

In these equations, V is the rod velocity, U is the penetration rate, P is the rod penetration, and L is the rod length, all functions of time. The constant parameters include the rod strength Y , the target resistance R , and the target-to-rod density ratio $\gamma = \rho_T/\rho_R$. The dots signify time differentiation. These equations have typically been integrated numerically to achieve a solution. A decade ago, Walters and Segletes [3] obtained an exact solution to these equations. However, the solution was not expressed in terms of the primitive variables that appear in the original equations, but rather in terms of an oblique transformation variable that was presented without explanation. Furthermore, little attempt was made to collate variables into an orderly fashion, thus leaving an incomplete sense for the term groupings that actually drive the solution. While mathematically rigorous, the solution was somewhat cumbersome to use.

This equation set has been re-examined, in search of improvements and extensions to the solution method. Several improvements to the solution approach are offered herein to improve the solution efficiency. A primary hindrance of the original solution was in the evaluation of the rod velocity as a function of time. While this hindrance remains with the current approach, it may be circumvented by choosing an independent variable other than time, in the evaluation of rod erosion. Indeed, it is often more useful to express the solution in terms of, for example, rod velocity, rather than the canonical function-of-time solution. And while this alternative was available to the original solution [3], the presentation of the original solution perhaps left the false impression with the reader that the numerical evaluation of $V(t)$ was a necessary intermediate step in the solution of $L(V)$ and $P(V)$.

Though the governing equations (1)–(4) pertain only to the time during which penetration and erosion simultaneously occur, extensions to the original solution [3] are herein provided for the subsequent stage of rigid-body penetration or rigid-target rod erosion. In addition to the general-case solution to the penetration problem being addressed, several special-case conditions, including the cases for which $R = Y$ and $\rho_R = \rho_T$, respectively, will also be solved. Not addressed

herein, however, because of their simplicity, are three special cases for which $R = 0$, $Y = 0$, and $R = Y = 0$, respectively. The present method, described subsequently, can be used to describe the $R = 0$ solution up until the moment that rigid-body penetration commences. Subsequent behavior, however, will be governed by Poncelet flow. In the case of both $Y = 0$ and $R = Y = 0$, the solution becomes trivial in that the rod velocity remains constant until the rod is totally consumed, at which point the event ceases. The penetration velocity and rod erosion rate also remain constant for these cases, in accordance with equations (2) and (3). For the case of $R = Y = 0$, the steady-state erosion rates are governed by the Bernoulli equation.

2. Closed Form Solution for $L(V)$

Without delay, we present the solution to the rod erosion equations, which is valid for all cases (special cases [$R = Y$, $\rho_R = \rho_T$] and the general case):

$$\frac{L}{L_0} = \left(\frac{\sqrt{\gamma}U - \dot{L}}{\sqrt{\gamma}U_0 - \dot{L}_0} \right)^{\frac{1}{\sqrt{\gamma}}\left(\frac{R}{Y}-1\right)} \exp\left[\frac{V_0\dot{L}_0 - V\dot{L}}{2Y/\rho_R} \right], \quad (5)$$

where the “0” subscripts signify conditions at the onset of the penetration event. It is worthy to note that while $-\dot{L}$ is the rate of rod erosion, the term $\sqrt{\gamma}U$ would be the rate of rod erosion were the case hydrodynamic (i.e., where $R=Y=0$).

The presentation of the solution, given by equation (5), is obtained by solving equation (1) for L and differentiating with respect to time; then, using equations (2) and (3) to obtain dL/dt in terms of rod velocity V ; and finally eliminating dL/dt from the two resulting equations, which gives d^2V/dt^2 in terms of dV/dt and V . The particular expression for dL/dt varies depending on whether the special or general problem cases are considered, and this will affect the form of the governing equation, as will be shown. The resulting equation is integrated to provide dV/dt in terms of V . While the traditional technique is to separate variables and attempt to integrate again to obtain $V(t)$, as was done by Walters and Segletes [3], this step is not necessary to obtain $L(V)$. Equation (1) provides a direct algebraic link between L and dV/dt , and thereby allows dV/dt to be eliminated in favor of L , immediately following the first integration. The result is $L(V)$, which is a desirable way to express the result, as an alternative to $L(t)$.

When special- and general-case problems are considered, the solutions, at first glance, take on different appearances. However, equation (5) was discerned from those solutions (given here, expressed in terms of a single independent variable, V , the rod velocity) by realizing that the various grouping of V terms in the various $L(V)$ solutions all satisfy the elegant form of equation (5):

$R = Y$:

$$\text{Governing Equation: } \frac{\sqrt{\gamma}}{1 + \sqrt{\gamma}} V \dot{V}^2 = -\frac{Y}{\rho_R} \ddot{V}, \quad (6)$$

$$\frac{L}{L_0} = \exp\left[\frac{-\rho_R \sqrt{\gamma}}{2Y(1 + \sqrt{\gamma})} (V_0^2 - V^2)\right], \quad (7)$$

$\gamma = 1$:

$$\text{Governing Equation: } \left(\frac{V}{2} + \frac{R - Y}{\rho_R V}\right) \dot{V}^2 = -\frac{Y}{\rho_R} \ddot{V}, \quad (8)$$

$$\frac{L}{L_0} = \left(\frac{V}{V_0}\right)^{\left(\frac{R}{Y} - 1\right)} \exp\left[\frac{-\rho_R}{4Y} (V_0^2 - V^2)\right], \quad (9)$$

General case:

$$\text{Governing Equation: } \frac{1}{1 - \gamma} \left(-\gamma V + \sqrt{\gamma V^2 + 2(R - Y)(1 - \gamma)/\rho_R}\right) \dot{V}^2 = -\frac{Y}{\rho_R} \ddot{V}, \quad (10)$$

$$\begin{aligned} \frac{L}{L_0} = & \left(\frac{V}{V_0} \cdot \frac{1 + \sqrt{1 + 2(R - Y)(1 - \gamma)/(\gamma \rho_R V^2)}}{1 + \sqrt{1 + 2(R - Y)(1 - \gamma)/(\gamma \rho_R V_0^2)}}\right)^{\frac{1}{\sqrt{\gamma}}\left(\frac{R}{Y} - 1\right)} \cdot \exp\left[\frac{-\rho_R \sqrt{\gamma}}{2Y(1 + \sqrt{\gamma})} \times \right. \\ & \left. \left(V_0^2 \cdot \frac{\sqrt{1 + 2(R - Y)(1 - \gamma)/(\gamma \rho_R V_0^2)} - \sqrt{\gamma}}{1 - \sqrt{\gamma}} - V^2 \cdot \frac{\sqrt{1 + 2(R - Y)(1 - \gamma)/(\gamma \rho_R V^2)} - \sqrt{\gamma}}{1 - \sqrt{\gamma}}\right)\right]. \end{aligned} \quad (11)$$

Equations (7), (9), and (11) have been organized and presented in a manner to demonstrate the functional linkage between the special- and general-case solutions. For example, when either $R = Y$ or $\gamma = 1$, the extended square-root terms of equation (11) become unity, leading to the simpler (V/V_0) monomial and $(V_0^2 - V^2)$ exponential terms of equations (9) and (7). When $\gamma = 1$, the leading multiplier on the exponential term in equation (11) matches that of equation (9). And when $R = Y$, the exponent on the monomial becomes zero, leading to the form of equation (7). While the forms for $U(V)$ and $\dot{L}(V)$, obtainable from equations (2) and/or (3), are vastly different in appearance for the special and general cases, the solutions for $L(V)$ nonetheless all share a common structured form described by equation (5).

3. Choice of Model Variable

While equations (6), (8) and (10) of the previous section choose to cast the problem in terms of rod velocity and its derivatives, this is by no means the only option. Through equations (2) and (3), V may be algebraically expressed in terms of U or \dot{L} . Therefore, instead of expressing rod length as $L = L(V)$ in equation (11), alternate expressions of the result, given as $L = L(U)$ or $L = L(\dot{L})$ may be obtained as:

$$\frac{L}{L_0} = \left(\frac{U}{U_0} \cdot \frac{1 + \sqrt{1 + 2(R-Y)/(\gamma \rho_R U^2)}}{1 + \sqrt{1 + 2(R-Y)/(\gamma \rho_R U_0^2)}} \right)^{\frac{1}{\sqrt{\gamma}} \left(\frac{R}{Y} - 1 \right)} \cdot \exp \left[\frac{-\rho_R \sqrt{\gamma} (1 + \sqrt{\gamma})}{2Y} \times \right. \\ \left. \left(U_0^2 \cdot \frac{\sqrt{1 + 2(R-Y)/(\gamma \rho_R U_0^2)} + \sqrt{\gamma}}{1 + \sqrt{\gamma}} - U^2 \cdot \frac{\sqrt{1 + 2(R-Y)/(\gamma \rho_R U^2)} + \sqrt{\gamma}}{1 + \sqrt{\gamma}} \right) \right], \quad (12)$$

$$\frac{L}{L_0} = \left(\frac{\dot{L}}{\dot{L}_0} \cdot \frac{1 + \sqrt{1 - 2(R-Y)/(\rho_R \dot{L}^2)}}{1 + \sqrt{1 - 2(R-Y)/(\rho_R \dot{L}_0^2)}} \right)^{\frac{1}{\sqrt{\gamma}} \left(\frac{R}{Y} - 1 \right)} \cdot \exp \left[\frac{-\rho_R (1 + \sqrt{\gamma})}{2Y \sqrt{\gamma}} \times \right. \\ \left. \left(\dot{L}_0^2 \cdot \frac{\sqrt{1 - 2(R-Y)/(\rho_R \dot{L}_0^2)} + \sqrt{\gamma}}{1 + \sqrt{\gamma}} - \dot{L}^2 \cdot \frac{\sqrt{1 - 2(R-Y)/(\rho_R \dot{L}^2)} + \sqrt{\gamma}}{1 + \sqrt{\gamma}} \right) \right]. \quad (13)$$

In the derivation of these and subsequent relations, there are several closely related, algebraic expressions that can facilitate expression and/or transformation of results. These include:

$$\sqrt{\gamma} U - \dot{L} = \left[\sqrt{\gamma} V + \sqrt{\gamma V^2 + 2(1-\gamma)(R-Y)/\rho_R} \right] / (1 + \sqrt{\gamma}); \quad (14)$$

$$\sqrt{\gamma} U + \dot{L} = \left[\sqrt{\gamma} V - \sqrt{\gamma V^2 + 2(1-\gamma)(R-Y)/\rho_R} \right] / (1 - \sqrt{\gamma}); \quad (15)$$

$$\dot{L} = \left[\gamma V - \sqrt{\gamma V^2 + 2(1-\gamma)(R-Y)/\rho_R} \right] / (1 - \gamma); \quad (16)$$

$$U = \left[V - \sqrt{\gamma V^2 + 2(1-\gamma)(R-Y)/\rho_R} \right] / (1 - \gamma). \quad (17)$$

4. Model-Variable Transformation

The complications of having the model variable V , U , or \dot{L} under the square root for the general case of equation (11), (12), or (13), respectively, may be circumvented with the selection of a mathematically more “natural” variable than the velocity V , U , or \dot{L} . Looking to equation (5) for guidance, success has been found in

$$\Phi = \frac{\sqrt{\gamma}U - \dot{L}}{\sqrt{|\Sigma|}}, \quad (18)$$

where the constant Σ is defined as $2(R - Y)/\rho_R$. The variable Φ , proportional to the expression of equation (14), is always nonnegative and follows somewhat the trend of rod velocity V (it actually equals $V/\sqrt{|\Sigma|}$ when $\gamma = 1$). Not surprisingly, Φ is also proportional to \sqrt{z} , which was the key transformation variable employed in the original derivation [3]. The key benefit to using the Φ transformation is that \dot{L} and U , rather than requiring square root terms as did equations (16) and (17) when expressed in V , may be expressed in more simply in terms of Φ as

$$\dot{L} = -\frac{\sqrt{|\Sigma|}}{2} \left(\Phi + \text{sgn}(\Sigma) \frac{1}{\Phi} \right), \quad (19)$$

and

$$U = \frac{\sqrt{|\Sigma|}}{2\sqrt{\gamma}} \left(\Phi - \text{sgn}(\Sigma) \frac{1}{\Phi} \right), \quad (20)$$

where the signum function, $\text{sgn}(x)$, denotes the sign of the argument [$\text{sgn}(x) = x/|x|$ for $x \neq 0$, and $\text{sgn}(x) = 0$ for $x = 0$], in this case the sign of Σ . The rod velocity, V , may also be obtained directly, by substituting these expressions into equation (3):

$$V = \frac{\sqrt{|\Sigma|}}{2\sqrt{\gamma}} \left((\sqrt{\gamma} + 1)\Phi + \text{sgn}(\Sigma) \frac{(\sqrt{\gamma} - 1)}{\Phi} \right). \quad (21)$$

When Φ is used in preference to rod velocity V as the independent variable, the governing equation (5) leads to the following expression:

$$\frac{L}{L_0} = \left(\frac{\Phi}{\Phi_0} \right)^{\frac{1}{\sqrt{\gamma}} \left(\frac{R}{Y} - 1 \right)} \exp \left\{ -\frac{1}{4\sqrt{\gamma}} \left| \frac{R}{Y} - 1 \right| \left[\left((\sqrt{\gamma} + 1)\Phi_0^2 + \frac{(\sqrt{\gamma} - 1)}{\Phi_0^2} \right) - \left((\sqrt{\gamma} + 1)\Phi^2 + \frac{(\sqrt{\gamma} - 1)}{\Phi^2} \right) \right] \right\}. \quad (22)$$

With minimal rearrangement, the variable Φ can be made to appear always in squared form. It is for this reason that Walters and Segletes [3] selected their transformation variable, z ,

proportional to Φ^2 . We will do the same here, though with a different proportionality constant, so that

$$z = \Phi^2 \cdot \sqrt{\frac{\sqrt{\gamma} + 1}{|\sqrt{\gamma} - 1|}}. \quad (23)$$

By so doing, the expression for residual rod length, equation (22), becomes

$$\frac{L}{L_0} = (z/z_0)^{\frac{1}{2\sqrt{\gamma}}\left(\frac{R}{Y}-1\right)} \exp\left\{-\frac{\sqrt{|\gamma-1|}}{4\sqrt{\gamma}}\left|\frac{R}{Y}-1\right|[(z_0 \pm 1/z_0) - (z \pm 1/z)]\right\}, \quad (24)$$

where the conditional operators in equation (24) are chosen as “+” for $\gamma > 1$ and “-” for $\gamma < 1$, and

$$\sqrt{z} = \left(\frac{\sqrt{\gamma} + 1}{|\sqrt{\gamma} - 1|\Sigma^2}\right)^{1/4} \cdot (\sqrt{\gamma}U - \dot{L}) = \frac{\sqrt{\gamma}V + \sqrt{\gamma}V^2 + (1-\gamma)\Sigma}{\left[|\sqrt{\gamma} - 1|(\sqrt{\gamma} + 1)^3 \Sigma^2\right]^{1/4}}. \quad (25)$$

Like equation (11), the result given by equation (24) expresses rod length in terms of a single independent variable, in this case z . The advantage of equation (24) over equation (11) is in removing the model variable from under a radical. The choice of a proportionality constant different than that used in the prior work [3], when defining z , provides a result that reduces the number of constant parameters in the exponent. More importantly, however, the appearance in the exponential of the model variable in the specific form of $(z \pm 1/z)$ will greatly expedite the evaluation of rod penetration, as will be subsequently shown.

5. Penetration

The evaluation of penetration by way of integrating equation (4) may be transformed with equation (1) to give

$$P = \int_0^t U dt = -\frac{1}{V_0} \int_V^{V_0} \frac{L}{L_0} U dV. \quad (26)$$

The particular functional forms for L and U will govern the form of the solution.

5.1 $R = Y$

Penetration may be directly evaluated in closed form for the simple case of $R=Y$, wherein L is given by equation (7), and U is proportional to V throughout the penetration event. In this case,

the final penetration (given by P_f as U and V approach zero) is

$$R = Y: P_f = \frac{L_0}{\sqrt{\gamma}} \left(1 - \exp \left[\frac{-\rho_R \sqrt{\gamma}}{2Y(1 + \sqrt{\gamma})} V_0^2 \right] \right). \quad (27)$$

5.2 $\gamma = 1$

For the $\gamma = 1$ special case, where the penetration velocity U is given algebraically by $U = (V - \Sigma/V)/2$, the penetration may be calculated, per equation (26), in closed form if the value of the V exponent in equation (9), given as $(R - Y)/Y$, is an even integer (i.e., R/Y is an odd integer). For these limited cases, integration by parts permits the problem reduction according to one of the following two recursion rules:

$$\int (V^2)^a e^{bV^2} dV^2 = \frac{1}{b} (V^2)^a e^{bV^2} - \frac{a}{b} \int (V^2)^{a-1} e^{bV^2} dV^2, \quad (a > 0); \quad (28a)$$

$$\int (V^2)^a e^{bV^2} dV^2 = \frac{1}{(a+1)} (V^2)^{a+1} e^{bV^2} - \frac{b}{(a+1)} \int (V^2)^{a+1} e^{bV^2} dV^2, \quad (a < 0); \quad (28b)$$

which is repeated until the reduced exponent on V^2 becomes zero, whereupon the process terminates with the rule

$$\int e^{bV^2} dV^2 = e^{bV^2} / b. \quad (29)$$

For other cases without the appropriate integral exponents, a recursion-type solution, with tabulation is still plausible, in theory. The recursion rule would be applied j times until the exponent $a \pm j$ falls between zero and unity. At that point, a tabulated solution for values of a between zero and unity (along the lines of the Gamma-function solution) is used to close out the integration. With clever use of velocity-normalization, the integral from V_0^2 to V_x^2 can be broken into two integrals, each evaluated between 0 and 1. Unfortunately, the integral remains a function of two parameters, a and b . And while tabulating a function of one parameter can be an efficient solution technique, tabulating solutions for functions of two real parameters quickly become more cumbersome than a series expansion or numerically integrated solution.

As an alternative then, the penetration for the $\gamma = 1$ special case may be evaluated by way of series solution in terms of velocity. One way to achieve this is to express the penetration as

$$\gamma = 1: \frac{P}{L_0} = \left[\sum_{j=0}^{\infty} a_j \left(\frac{\rho_R V_0^2}{4Y} \right)^j \right] - \frac{L}{L_0} \left[\sum_{j=0}^{\infty} a_j \left(\frac{\rho_R V^2}{4Y} \right)^j \right], \quad (30)$$

and match the derivative of P to the terms of U , given by $U = (V - \Sigma/V)/2$. With this approach, one obtains $a_0 = -1$, $a_1 = 2/(1 + \rho_R \Sigma/4Y)$, and for the remaining terms, $a_j = -a_{j-1}/(j + \rho_R \Sigma/4Y)$. Note that $\rho_R \Sigma/4Y$ equals $(R/Y - 1)/2$. While the series terms alternate in sign, the fact that j is in

the denominator of the recursion formula indicates that the rate of convergence for this solution approach should be similar to that for the exponential series. To confirm that this expression approaches the proper form for the special case when $R=Y$ (when Σ equals zero), the recursion relation is observed to then approach the series definition for the exponential, $1 - 2 \exp[-V^2/4K]$. This series takes on the value $1 - 2(L_f/L_0)$ when evaluated at V_0 , and -1 when evaluated at $V=0$. Here, L_f is the terminal length of the rod. The final penetration, therefore, becomes $L_0 - L_f$, as expected for $R=Y$ and $\gamma=1$.

Perhaps a more forthright approach for the evaluation of penetration for the $\gamma=1$ special case (and less prone to the precision problems of evaluating an alternating series) is to directly integrate $LU \cdot dV$, per equation (26), by initially expanding the exponential term of L into a series,

$$\frac{L}{L_0} \cdot U = \frac{1}{2} \exp\left(\frac{-\rho_R V_0^2}{4Y}\right) \left(V - \frac{\Sigma}{V}\right) \left(\frac{V}{V_0}\right)^{\frac{\rho_R \Sigma}{2Y}} \left(1 + \frac{(\rho_R V^2/4Y)}{1!} + \frac{(\rho_R V^2/4Y)^2}{2!} + \dots\right), \quad (31)$$

and integrating term by term to the desired level of precision. As before, $\Sigma = 2(R - Y)/\rho_R$. By integrating this expression with respect to V , per equation (26), one may obtain

$\gamma = 1$:

$$\frac{P}{L_0} = \exp\left[\frac{-\rho_R V_0^2}{4Y}\right] \left\{ \sum_{j=0}^{\infty} \frac{1}{j!} \cdot \frac{j - \frac{\rho_R \Sigma}{4Y}}{j + \frac{\rho_R \Sigma}{4Y}} \cdot \left(\frac{\rho_R V_0^2}{4Y}\right)^j - \left(\frac{V^2}{V_0^2}\right)^{\frac{\rho_R \Sigma}{4Y}} \sum_{j=0}^{\infty} \frac{1}{j!} \cdot \frac{j - \frac{\rho_R \Sigma}{4Y}}{j + \frac{\rho_R \Sigma}{4Y}} \cdot \left(\frac{\rho_R V^2}{4Y}\right)^j \right\}. \quad (32)$$

5.3 General Case

In evaluating the penetration for the general case, the solution becomes more complicated but can nonetheless be made more efficient compared to the method presented in the original solution [3]. Efficiencies are achieved in several ways. The use of rod length L in the form of equation (24) retains integer-powered polynomials in the exponential term. As such, the series expansion of the exponential, by which the integrals are evaluated, does not require the evaluation of fractionally powered polynomial expansions, as did the original method [3]. But more importantly, by having transformed L into a form where the exponential argument is of the explicit form $c(z \pm 1/z)$, a method may be used to expand the exponential in an efficient way, reducing the expansion of the exponential to power n from a cost of $(n+1)(n+2)/2$ monomial evaluations in z , to one of $2n+1$ evaluations in z .

The equation describing the penetration, equation (26), may be reorganized to obtain an expression in terms of the transformation variable, z ,

$$P = \int_0^t U dt = \int_{z_0}^z \frac{U}{\dot{V}} \frac{dV}{dz} dz = -\frac{1}{\dot{V}_0} \int_z^{z_0} \frac{L}{L_0} U \frac{dV}{dz} dz. \quad (33)$$

Using equations (21) and (23), the rod velocity is expressible in terms of z as

$$V = \frac{(|\gamma - 1|\Sigma^2)^{1/4}}{2\sqrt{\gamma}} \left[(\sqrt{\gamma} + 1)^{1/2} \sqrt{z} + \text{sgn}[(\gamma - 1)\Sigma] \frac{|\sqrt{\gamma} - 1|^{1/2}}{\sqrt{z}} \right], \quad (34)$$

so that dV/dz may be computed as

$$\frac{dV}{dz} = \frac{(|\gamma - 1|\Sigma^2)^{1/4}}{4\sqrt{\gamma}} \left[\frac{(\sqrt{\gamma} + 1)^{1/2}}{z^{1/2}} - \text{sgn}[(\gamma - 1)\Sigma] \frac{|\sqrt{\gamma} - 1|^{1/2}}{z^{3/2}} \right]. \quad (35)$$

In a similar vein, from equations (20) and (23), U may be expressed in terms of z as

$$U = \frac{\sqrt{|\Sigma|}}{2\sqrt{\gamma}} \left[\left(\frac{|\sqrt{\gamma} - 1|}{\sqrt{\gamma} + 1} \right)^{1/4} \sqrt{z} - \text{sgn}(\Sigma) \left(\frac{\sqrt{\gamma} + 1}{|\sqrt{\gamma} - 1|} \right)^{1/4} \frac{1}{\sqrt{z}} \right]. \quad (36)$$

The product, $U \cdot dV/dz$, may therefore be computed as

$$U \frac{dV}{dz} = \frac{|\gamma - 1|^{1/4} |\Sigma|}{8\gamma} \times \left\{ |\gamma - 1|^{1/4} - \text{sgn}(\Sigma) \left[\left[\frac{(\sqrt{\gamma} + 1)^3}{|\sqrt{\gamma} - 1|} \right]^{1/4} + \text{sgn}(\gamma - 1) \left[\frac{|\sqrt{\gamma} - 1|^3}{(\sqrt{\gamma} + 1)} \right]^{1/4} \right] \frac{1}{z} + \text{sgn}(\gamma - 1) \frac{|\gamma - 1|^{1/4}}{z^2} \right\}, \quad (37)$$

which is of the form

$$U \cdot dV/dz = A(a_0 + a_1/z + a_2/z^2). \quad (38)$$

Substituting this result and the transformed expression for L , given by equation (24), into equation (33) allows the integral for penetration to take the form

$$P = B_p \int_z^{z_0} (a_0 + a_1/z + a_2/z^2) z^b \exp[c(z \pm 1/z)] dz, \quad (39)$$

where the conditional minus sign in the exponential is taken when $\gamma < 1$, and a_i , b , c , and B_p are all constants, expressible as

$$a_0 = |\gamma - 1|^{1/4}, \quad (40)$$

$$a_1 = -\operatorname{sgn}(R - Y) \left(\left[\frac{(\sqrt{\gamma}+1)^3}{|\sqrt{\gamma}-1|} \right]^{1/4} + \operatorname{sgn}(\gamma - 1) \left[\frac{|\sqrt{\gamma}-1|^3}{(\sqrt{\gamma}+1)} \right]^{1/4} \right), \quad (41)$$

$$a_2 = \operatorname{sgn}(\gamma - 1) |\gamma - 1|^{1/4}, \quad (42)$$

$$b = \frac{1}{2\sqrt{\gamma}} \left(\frac{R}{Y} - 1 \right), \quad (43)$$

$$c = \frac{1}{4} \sqrt{\frac{|\gamma - 1|}{\gamma}} \left| \frac{R}{Y} - 1 \right|, \quad (44)$$

and

$$B_p = L_0 \left| \frac{R}{Y} - 1 \right| \frac{|\gamma - 1|^{1/4}}{4\gamma \cdot z_0^b} \exp[-c(z_0 + \operatorname{sgn}(\gamma - 1)/z_0)]. \quad (45)$$

The form of equation (39) is basically identical to an intermediate step of the original solution [3], though with differently defined constants. The prior work [3] opted to transform the equation again to eliminate the leading polynomials, but did so at the expense of introducing noninteger powers into the exponential term. Then, the penetration equation was solved by expanding the exponential into a power series of $(A_1 \cdot z^s + A_2 \cdot z^{-s})^j$ terms and expanding each $(A_1 \cdot z^s + A_2 \cdot z^{-s})^j$ term into $j + 1$ monomials, using a binomial expansion. The net result of the total expansion was that, to include terms out to a power of $j = n$, a total of $(n + 1)(n + 2)/2$ monomials was generated, and then integrated term by term. With n routinely exceeding 20 to obtain the desired precision, and approaching 100 for certain initial conditions, the computational burden was substantial, though still more efficient than a numerical integration of equations (1)–(4).

While the currently proposed method still relies on a series expansion of the exponential to perform the integration, a technique permits a streamlined method for achieving the expansion. In particular, a method exists to expand the subject exponential series with the form

$$\exp[c(z \pm 1/z)] = \sum_{j=-\infty}^{\infty} C_j^{\pm} z^j, \quad (46)$$

where the C_j^+ or C_j^- coefficients are a function only of the parameter c . In particular, the C_j^- constants are given by evaluations of Bessel functions of the first kind, such that $C_j^- = J_j(2c)$. The C_j^+ constants, by contrast, are given by modified Bessel functions of the first kind, such that $C_j^+ = I_j(2c)$. The expansion using the form of equation (46), to include terms of power $z^{\pm n}$,

requires the evaluation of only $2n + 1$ monomials in z , and therefore represents a significant improvement over the method previously employed [3], which required the evaluation of $(n + 1)(n + 2)/2$ monomials in z for identical precision.

While there is an overhead associated with the evaluation of the C_j^\pm parameters, given by the converging series that defines the Bessel functions for integer order,

$$C_j^\pm = \begin{cases} \sum_{i=0}^{\infty} \frac{(\pm 1)^i c^{2i+j}}{i!(i+j)!} & , j \geq 0 \\ (\pm 1)^j C_{-j}^\pm & , j < 0 \end{cases} , \quad (47)$$

the parameter c is fixed by the initial conditions (material properties) of the penetration problem. As such, the C_j^+ or C_j^- terms may be calculated once at the onset of the analysis, regardless of how many z values (i.e., velocities) for which the solution needs evaluation. Furthermore, there exists a recursive technique for evaluating the C_j^\pm parameters of equation (47), based on the recursions

$$\frac{C_j^+}{C_{j-1}^+} = \frac{1}{\frac{j}{c} + \frac{C_{j+1}^+}{C_j^+}} , \quad (48a)$$

and

$$\frac{C_j^-}{C_{j-1}^-} = \frac{1}{\frac{j}{c} - \frac{C_{j+1}^-}{C_j^-}} , \quad (48b)$$

which thereby offers further computational savings.

The integration for penetration is, thus, finally achieved by employing this optimized expansion and integrating term by term and evaluating at the desired limits. When b is not an integer, which is the typical case, the result may be expressed as

$$\text{General case: } P = B_p \sum_{j=-\infty}^{\infty} (a_0 C_{j-1}^\pm + a_1 C_j^\pm + a_2 C_{j+1}^\pm) \frac{z^{j+b}}{j+b} \Bigg|_z^{z_0} , \quad (49)$$

where the C^+ terms are used when $\gamma > 1$ and the C^- terms are used when $\gamma < 1$. For the case when b is an integer, the single term of equation (49) that would otherwise produce a zero in the denominator (i.e., the term for which $j = -b$) originated from a $1/z$ integration and would actually have produced, upon integration, the logarithmic term $\ln(z)$, instead of $z^{j+b}/(j+b)$.

6. Implicit Time

Though these solutions for $L(V)$ and $P(V)$ bypass the intermediate evaluation of $V(t)$, the penetration variables may, if needed, be implicitly expressed in terms of time, by integration of $L(V)$,

$$t = \int_{V_0}^V \frac{dV}{\dot{V}} = -\frac{1}{\dot{V}_0} \int_V^{V_0} \frac{L}{L_0} dV, \quad (50)$$

in order to obtain $t(V)$. As in the case of penetration, a closed-form solution to equation (50) will be possible only for the special case of $\gamma = 1$ and then only when $(R - Y)/Y$ is an odd integer (i.e., R/Y is even). In all other cases, the integration of equation (50) will take the form of a series solution. Of the several ways to obtain a series integration of the special case solutions, a power-series expansion is preferable to a repeated integration-by-parts solution because it avoids an alternating series, for the case when the “ c ” constant associated with the $\exp[-c(V_0^2 - V^2)]$ term is positive. Such is always the case for penetration problems. Both the $R = Y$ and $\gamma = 1$ special cases can be reduced to an integral of the form

$$\int_0^a V^b \exp(cV^2) dV = a^{b+1} \sum_{i=0}^{\infty} \frac{(ca^2)^i}{i!(2i+b+1)}. \quad (51)$$

Thus, the special-case solutions for $t(V)$ may be evaluated as

$R = Y$:

$$t = \frac{\rho_R L_0 V_0}{Y} \exp\left[\frac{-\rho_R \sqrt{\gamma} V_0^2}{2Y(1+\sqrt{\gamma})}\right] \left[\sum_{i=0}^{\infty} \frac{1}{i!(2i+1)} \left(\frac{\rho_R \sqrt{\gamma} V_0^2}{2Y(1+\sqrt{\gamma})}\right)^i - \frac{V}{V_0} \sum_{i=0}^{\infty} \frac{1}{i!(2i+1)} \left(\frac{\rho_R \sqrt{\gamma} V^2}{2Y(1+\sqrt{\gamma})}\right)^i \right], \quad (52)$$

$\gamma = 1$:

$$t = \frac{\rho_R L_0 V_0}{Y} \exp\left[\frac{-\rho_R V_0^2}{4Y}\right] \left[\sum_{i=0}^{\infty} \frac{1}{i!(2i+R/Y)} \left(\frac{\rho_R V_0^2}{4Y}\right)^i - \left(\frac{V}{V_0}\right)^{R/Y} \sum_{i=0}^{\infty} \frac{1}{i!(2i+R/Y)} \left(\frac{\rho_R V^2}{4Y}\right)^i \right]. \quad (53)$$

For the general case, a solution is most profitably obtained in a manner analogous to the penetration evaluation, in which a transformation to z facilitates a streamlined series solution:

$$t = \int_{V_0}^V \frac{dV}{\dot{V}} = -\frac{1}{\dot{V}_0} \int_z^{z_0} \frac{L}{L_0} \frac{dV}{dz} dz. \quad (54)$$

This integration may be staged through the substitution of equations (24) and (35), to give the following form:

$$t = B_t \int_z^{z_0} (d_0/z^{1/2} + d_1/z^{3/2}) z^b \exp[c(z \pm 1/z)] dz, \quad (55)$$

where the conditional minus sign is taken when $\gamma < 1$. Here, b and c are defined as before, by equations (43) and (44), while

$$d_0 = (\sqrt{\gamma} + 1)^{1/2}, \quad (56)$$

$$d_1 = -\text{sgn}[(\gamma - 1)(R - Y)] |\sqrt{\gamma} - 1|^{1/2}, \quad (57)$$

and

$$B_t = L_0 \sqrt{\frac{\rho_R}{Y}} \left(\frac{1}{8\gamma} \cdot \left| \frac{R}{Y} - 1 \right| \right)^{1/2} \frac{|\gamma - 1|^{1/4}}{z_0^b} \exp[-c(z_0 + \text{sgn}(\gamma - 1)/z_0)]. \quad (58)$$

By using a method analogous to that in equations (46)–(49) and with the same definitions for C_j^\pm [given by equation (47), where the “+” solution applies for $\gamma > 1$, and the “–” solution for $\gamma < 1$], the expression for t given by equation (55) may be expanded in a series as

$$\text{General case: } t = B_t \sum_{j=-\infty}^{\infty} (d_0 C_{j-1}^\pm + d_1 C_j^\pm) \frac{z^{j+b-1/2}}{j+b-1/2} \Bigg|_z^{z_0}. \quad (59)$$

Like equation (49), there is one exception to the general validity of this result, specifically for the case when b is precisely a half-integer. If and only if this is the case, a single term of equation (59) will require modification: namely, the term for which $j + b - 1/2$ exactly equals zero, originating from a $1/z$ integration. This integration would, for this one term only, rightfully have produced a $\ln(z)$ term, instead of $z^{j+b-1/2}/(j+b-1/2)$. As with the evaluation of penetration, the summation of equation (59) is carried out for j over some finite range from $-n$ to $+n$ so as to achieve the desired level of precision.

7. Terminal Rod Length, etc.

The “terminal” rod length may be ascertained for the various solution cases [from equations (7), (9) or (11)], by setting V to its terminal value, $V_x = \sqrt{\Sigma}$ for the case of $R > Y$ and $V_x = \sqrt{-\Sigma/\gamma}$ for $R < Y$, with the parameter Σ given by $\Sigma = 2(R - Y)/\rho_R$. When $R > Y$, this termination corresponds to the point where $U = 0$, when the penetration ceases (though the rod may continue to erode thereafter). For $R < Y$, the termination corresponds to the point where $\dot{L} = 0$, when the

rod erosion ceases (though the rod may continue to penetrate as a rigid body thereafter). This terminal state, denoted with the subscript “ x ,” corresponds not to the end of the ballistic event, but rather to the time at which the governing equations (1)–(4) cease to apply. In those governing equations, developed for the case of a simultaneously eroding rod and target, the subscript “ x ” condition corresponds to the moment at which either the rod or the target stops eroding. In general, these two conditions do not occur simultaneously. The rod length (normalized) at the terminal state “ x ” for the various cases is expressible as:

$$R=Y: \frac{L_x}{L_0} = \exp\left[\frac{-\rho_R \sqrt{\gamma}}{2Y(1+\sqrt{\gamma})} V_0^2\right], \quad (60)$$

$$\gamma = 1: \frac{L_x}{L_0} = \left[\frac{V_0^2}{|\Sigma|}\right]^{-\frac{1}{2}\left(\frac{R}{Y}-1\right)} \exp\left[-\frac{1}{2}\left|\frac{R}{Y}-1\right|\left(\frac{V_0^2}{|\Sigma|}-1\right)\right], \quad (61)$$

General case:

$$\frac{L_x}{L_0} = \left(\frac{\sqrt{\gamma}(V_0/\sqrt{|\Sigma|}) + \sqrt{\gamma(V_0/\sqrt{|\Sigma|})^2 + \text{sgn}(\Sigma)(1-\gamma)}}{1+\sqrt{\gamma}}\right)^{-\frac{1}{\sqrt{\gamma}}\left(\frac{R}{Y}-1\right)} \times \exp\left[-\left|\frac{R}{Y}-1\right|\left(\frac{(V_0/\sqrt{|\Sigma|})\sqrt{\gamma(V_0/\sqrt{|\Sigma|})^2 + \text{sgn}(\Sigma)(1-\gamma)} - \gamma(V_0/\sqrt{|\Sigma|})^2}{1-\gamma} - \frac{1+\text{sgn}(\Sigma)}{2}\right)\right]. \quad (62)$$

For cases where $R > Y$, this terminal length corresponds to that length of rod as of the moment that penetration ceases. For $R < Y$, this is the rod length at the onset of rigid-body penetration.

Terminal values (at state “ x ”) for penetration and time may likewise be obtained by evaluating the respective relations [equations (30), (32), or (49) for penetration and equations (52), (53), or (59) for time] with the substitution of $V = V_x$ [and $L = L_x$ in the case of equation (30)]. Their presentation is omitted, however, because these relations are summations and not in closed form like those for residual length previously given. As such, there is little clarity of reduction gained in restating these earlier equations with the $V = V_x$ substitution in place.

8. Residual Erosion/Penetration Behaviors

Equations (1) and (2) are valid only while there is simultaneous target penetration and rod erosion. Except for the special case of $R = Y$, \dot{L} and U will not simultaneously approach zero. In the general case then, the physical event will continue with either residual rod erosion

following the cessation of penetration (when $R > Y$) or residual rigid body penetration following the cessation of rod erosion (when $R < Y$). These afterflow events are amenable to closed-form analytical solution. Continuing to denote the state at this transition point (the moment of transition to either rigid target or rigid rod) with the use of the subscript “x,” the absolute final state, when the rod velocity itself finally reaches zero, will be denoted with the subscript “f.” Recall that $V_x = \sqrt{\Sigma}$ when $R > Y$, while $V_x = \sqrt{-\Sigma/\gamma}$ when $R < Y$, where $\Sigma = 2(R - Y)/\rho_R$.

8.1 Residual Rod Erosion

For the case of $R > Y$, the target becomes rigid while rod erosion continues. To deal with this, equation (2) is discarded and is substituted with the constraint $U = 0$. The kinematic constraint of equation (3) becomes, as a result, $V = -\dot{L}$. Equation (1) remains valid for the eroding-rod case. Solving equation (1) for L , differentiating, and substituting the revised kinematic constraint to eliminate \dot{L} , one obtains as the governing equation

$$V\dot{V}^2 = -(Y/\rho_R)\dot{V}^2. \quad (63)$$

This may be integrated to obtain \dot{V} in terms of V , whereupon equation (1) may be used to eliminate \dot{V} in favor of L . The result (as a function of V) is that

$$L = L_x \exp\left[\frac{-\rho_R}{2Y}(V_x^2 - V^2)\right]. \quad (64)$$

Evaluating the penetration and rod length at the final state (where $V = 0$), one obtains $P_f = P_x$ and

$$L_f = L_x \exp\left[\frac{-\rho_R V_x^2}{2Y}\right]. \quad (65)$$

Because of the similarity between the governing equation here, equation (63), and the special case $R = Y$ governing equation, equation (6), the duration of this residual-erosion phase of the rod may likewise be calculated with the same series-solution form used to calculate event duration for the special cases, described by equation (51). Use of this form leads to

$$t - t_x = \frac{\rho_R L_x V_x}{Y} \exp\left(\frac{-\rho_R V_x^2}{2Y}\right) \left[\sum_{i=0}^{\infty} \frac{1}{i!(2i+1)} \left(\frac{\rho_R V_x^2}{2Y}\right)^i - \frac{V}{V_x} \sum_{i=0}^{\infty} \frac{1}{i!(2i+1)} \left(\frac{\rho_R V^2}{2Y}\right)^i \right], \quad (66)$$

which, as V approaches zero, becomes the following result:

$$t_f - t_x = \frac{\rho_R L_x V_x}{Y} \exp\left(\frac{-\rho_R V_x^2}{2Y}\right) \sum_{i=0}^{\infty} \frac{1}{i!(2i+1)} \left(\frac{\rho_R V_x^2}{2Y}\right)^i. \quad (67)$$

8.2 Residual Rigid-Body Penetration

For the alternate case of $R < Y$, a state of rigid-body penetration is reached after the rod erosion ceases. As before, equation (2) is discarded and is substituted with the constraint $\dot{L} = 0$. The kinematic constraint (3) becomes, as a result, $V = U$. However, there is one additional modification required to the governing equations. In particular, the force causing the rod deceleration in equation (1) is no longer Y , since the rod is no longer in a plastic state. Rather, it is a diminished stress state applied by the pressure head and resistance of the target, $1/2\rho_T U^2 + R$. But since, kinematically, $V = U$ and L remains fixed at L_x , the rod deceleration equation becomes

$$L_x \dot{V} = -(1/2\rho_T V^2 + R) / \rho_R. \quad (68)$$

There is no algebraic relation between P and \dot{V} analogous to that which equation (1) affords between L and \dot{V} . Therefore, this equation will be solved by separating the variables V and t , as follows:

$$\frac{2L_x}{\gamma} \cdot \frac{dV}{2R / \rho_T + V^2} = -dt. \quad (69)$$

This may be solved as

$$V = U = \sqrt{\frac{2R}{\rho_T}} \tan \left[\frac{\gamma}{L_x} \sqrt{\frac{R}{2\rho_T}} (t_x - t) + \tan^{-1} \left(V_x \sqrt{\frac{\rho_T}{2R}} \right) \right]. \quad (70)$$

The final time, at which the velocity drops to zero, is found to be

$$t_f = t_x + \frac{L_x}{\gamma} \sqrt{\frac{2\rho_T}{R}} \tan^{-1} \left(V_x \sqrt{\frac{\rho_T}{2R}} \right). \quad (71)$$

The expression for U , which is equation (70), may be integrated one more time to obtain the differential penetration that occurs during the afterflow phase. One obtains

$$P - P_x = \frac{2L_x}{\gamma} \left\{ \log \cos \left[\frac{\gamma}{L_x} \sqrt{\frac{R}{2\rho_T}} (t_f - t) \right] - \log \cos \tan^{-1} \left(V_x \sqrt{\frac{\rho_T}{2R}} \right) \right\}. \quad (72)$$

When evaluated at $t = t_f$, and employing some trigonometric substitutions, the final result is that $L_f = L_x$ and the afterflow penetration is

$$P_f - P_x = \frac{L_x}{\gamma} \log \left(1 + \frac{\rho_T V_x^2}{2R} \right). \quad (73)$$

9. Conclusions

This report presents updated results related to the exact solution of the long-rod penetration equations, formulated by Alekseevskii [1] and Tate [2], and first solved by Walters and Segletes [3]. While the original solution [3] is accurate and comprehensive, there have been a number of improvements or enhancements, both to the presentation and the solution approach.

Equation (5) is a concise analytical presentation of rod length as a function of rod velocity, valid for both special and general cases, providing an enhanced sense for the terms that drive the analytical solution. Equations (6)–(11) compare and contrast the special- and general-case analytical solutions, while equations (12) and (13) present the result in terms of an alternate model variable. The key independent variable transformation (to z), unexplained but indispensable to the original solution, is herein developed more fully and much of its mystery is thereby uncloaked. Further, its expression is slightly altered from the original solution, resulting, by comparison, in a form amenable to a highly streamlined series solution for penetration $P(z)$, as equation (49), or implicit time, $t(z)$, as equation (59).

Not only are results derived to the point where the penetration equations cease validity, but extensions to the original solution are presented, which account for the period of rigid-body penetration or rigid-target rod erosion that follows the period of eroding-body penetration addressed by the original penetration equations.

While not taking anything from the original solution of Walters and Segletes [3], the current work offers enhanced appreciation and understanding of the original effort, as well as extensions to the original work. Finally, the streamlined techniques presented herein make any implementation of the solution significantly more efficient than the originally offered solution technique.

10. References

1. Alekseevskii, V. P. “Penetration of a Rod Into a Target at High Velocity.” *Combustion Explosion and Shock Waves*, vol. 2, pp. 63–66, 1966.
2. Tate, A. “A Theory for the Deceleration of Long Rods After Impact.” *Journal of the Mechanics and Physics of Solids*, vol. 15, pp. 387–399, 1967.
3. Walters, W. P., and S. B. Segletes. “An Exact Solution of the Long Rod Penetration Equations.” *International Journal of Impact Engineering*, vol. 11, no. 2, pp. 225–231, 1991.

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Service, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.			
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE September 2002	3. REPORT TYPE AND DATES COVERED Final, September 2001 - September 2002	
4. TITLE AND SUBTITLE Efficient Solution of the Long-Rod Penetration Equations of Alekseevskii-Tate		5. FUNDING NUMBERS 1L162618AH80	
6. AUTHOR(S) Steven B. Segletes and William P. Walters			
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) U.S. Army Research Directorate ATTN: AMSRL-WM-TD Aberdeen Proving Ground, MD 21005-5066		8. PERFORMING ORGANIZATION REPORT NUMBER ARL-TR-2855	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)		10. SPONSORING/MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES			
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.		12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) The exact solution to the long-rod penetration equations is revisited, in search of improvements to the solution efficiency, while simultaneously enhancing the understanding of the physical parameters that drive the solution. Substantial improvements are offered in these areas. The presentation of the solution is simplified in a way that more tightly unifies the special- and general-case solutions to the problem. Added computational efficiencies are obtained by expressing the general-case solution for penetration and implicit time in terms of a series of Bessel functions. Other extensions and efficiencies are addressed, as well.			
14. SUBJECT TERMS long rods, penetration, erosion, exact solution, closed-form solution		15. NUMBER OF PAGES 21	
		16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT UNCLASSIFIED	18. SECURITY CLASSIFICATION OF THIS PAGE UNCLASSIFIED	19. SECURITY CLASSIFICATION OF ABSTRACT UNCLASSIFIED	20. LIMITATION OF ABSTRACT UL

NSN 7540-01-280-6500

19

Standard Form 298 (Rev. 2-89)
Prescribed by ANSI Std. Z39-18 298-102

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	DEFENSE TECHNICAL INFORMATION CENTER DTIC OCA 8725 JOHN J KINGMAN RD STE 0944 FT BELVOIR VA 22060-6218		<u>ABERDEEN PROVING GROUND</u>
		2	DIR USARL AMSRL CI LP (BLDG 305)
1	COMMANDING GENERAL US ARMY MATERIEL CMD AMCRDA TF 5001 EISENHOWER AVE ALEXANDRIA VA 22333-0001		
1	INST FOR ADVNCD TCHNLGY THE UNIV OF TEXAS AT AUSTIN 3925 W BRAKER LN STE 400 AUSTIN TX 78759-5316		
1	US MILITARY ACADEMY MATH SCI CTR EXCELLENCE MADN MATH THAYER HALL WEST POINT NY 10996-1786		
1	DIRECTOR US ARMY RESEARCH LAB AMSRL D DR D SMITH 2800 POWDER MILL RD ADELPHI MD 20783-1197		
1	DIRECTOR US ARMY RESEARCH LAB AMSRL CI AI R 2800 POWDER MILL RD ADELPHI MD 20783-1197		
3	DIRECTOR US ARMY RESEARCH LAB AMSRL CI LL 2800 POWDER MILL RD ADELPHI MD 20783-1197		
3	DIRECTOR US ARMY RESEARCH LAB AMSRL CI IS T 2800 POWDER MILL RD ADELPHI MD 20783-1197		

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
5	DEFENSE NUCLEAR AGENCY MAJ J LYON CDR K W HUNTER T FREDERICKSON R J LAWRENCE SPSP K KIBONG 6801 TELEGRAPH RD ALEXANDRIA VA 22310-3398	1	NAVAL AIR WARFARE CTR S A FINNEGAN BOX 1018 RIDGECREST CA 93556
2	COMMANDER US ARMY ARDEC AMSTA AR FSA E W P DUNN E BAKER PICATINNY ARSENAL NJ 07806-5000	4	COMMANDER NAVAL WEAPONS CENTER N FASIG CODE 3261 T T YEE CODE 3263 D THOMPSON CODE 3268 W J MCCARTER CODE 6214 CHINA LAKE CA 93555
1	COMMANDER US ARMY ARDEC AMSTA AR CCH V M D NICOLICH PICATINNY ARSENAL NJ 07806-5000	12	COMMANDER NAVAL SURFACE WARFARE CTR DAHLGREN DIVISION H CHEN D L DICKINSON CODE G24 C R ELLINGTON C R GARRETT CODE G22 W HOLT CODE G22 W E HOYE G22 R MCKEOWN J M NELSON M J SILL CODE H11 W J STROTHER A B WARDLAW JR L F WILLIAMS CODE G33 17320 DAHLGREN RD DAHLGREN VA 22448
1	COMMANDER US ARMY ARDEC E ANDRICOPOULOS PICATINNY ARSENAL NJ 07806-5000		
1	COMMANDER USA STRATEGIC DEFNS CMD CSSD H LL T CROWLES HUNTSVILLE AL 35807-3801		
4	COMMANDER US ARMY AVIATION & MISSILE CMD AMSAM RD PS WF S HILL D LOVELACE M SCHEXNAYDER G SNYDER REDSTONE ARSENAL AL 35898-5247	2	AIR FORCE ARMAMANENT LAB AFATL DLJR J FOSTER D LAMBERT EGLIN AFB FL 32542-6810
1	COMMANDER US ARMY AVIATION & MISSILE CMD AMSAM RD SS AA J BILLINGSLEY REDSTONE ARSENAL AL 35898	1	USAF PHILLIPS LABORATORY VTSI ROBERT ROYBAL KIRTLAND AFB NM 87117-7345
1	MIS DEFNS & SPACE TECHNOLOGY CSSD SD T K H JORDAN PO BOX 1500 HUNTSVILLE AL 34807-3801	2	USAF PHILLIPS LABORATORY PL WSCD F ALLAHDADI PV VTA D SPENCER 3550 ABERDEEN AVE SE KIRTLAND AFB NM 87117-5776
3	COMMANDER US ARMY RESEARCH OFFICE K IYER J BAILEY S F DAVIS PO BOX 12211 RTP NC 27709-2211	1	AFIT ENC D A FULK WRIGHT PATTERSON AFB OH 45433
		1	FEDERAL BUREAU OF INVESTIGATION FBI LABORATORY EXPLOSIVES UNIT M G LEONE 935 PENNSYLVANIA AVE NW WASHINGTON DC 20535

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	DIRECTOR LLNL F A HANDLER L182 PO BOX 808 LIVERMORE CA 94550	3	NASA JOHNSON SPACE CENTER E CHRISTIANSEN J L CREWS F HORZ MAIL CODE SN3 2101 NASA RD 1 HOUSTON TX 77058
1	DIRECTOR LLNL MS L282 W TAO PO BOX 808 LIVERMORE CA 94550	1	APPLIED RESEARCH LAB J A COOK 10000 BURNETT ROAD AUSTIN TX 78758
2	DIRECTOR LLNL MS L290 A HOLT J E REAUGH PO BOX 808 LIVERMORE CA 94550	5	JET PROPULSION LABORATORY IMPACT PHYSICS GROUP Z SEKANINA P WEISSMAN B WEST J ZWISSLER M ADAMS 4800 OAK GROVE DR PASADENA CA 91109
1	DIRECTOR LLNL W J NELLIS L299 PO BOX 808 LIVERMORE CA 94550	2	CALTECH J SHEPHERD MS 105 50 A P INGERSOLL MS 170 25 1201 E CALIFORNIA BLVD PASADENA CA 91125
1	DIRECTOR LLNL S G COCHRAN L389 PO BOX 808 LIVERMORE CA 94550	1	CALTECH G ORTON MS 169 237 4800 OAK GROVE DR PASADENA CA 91007
2	DIRECTOR LLNL MS L495 D GAVEL J HUNTER PO BOX 808 LIVERMORE CA 94550	1	DREXEL UNIVERSITY MEM DEPT 32ND & CHESTNUT ST PHILADELPHIA PA 19104
1	DIRECTOR LLNL R M KUKLO L874 PO BOX 808 LIVERMORE CA 94550	1	GEORGIA INSTITUTE OF TECHNOLOGY COMPUTATIONAL MODELING CENTER S ATLURI ATLANTA GA 30332-0356
4	ENERGETIC MATERIALS RSCH TSTNG CTR NEW MEXICO TECH D J CHAVEZ L LIBERSKY F SANDSTROM M STANLEY CAMPUS STATION SOCORRO NM 87801	1	JOHNS HOPKINS UNIVERSITY MAT SCI & ENGNNG DEPT M LI 102 MARYLAND HALL 3400 N CHARLES ST BALTIMORE MD 21218-2689

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
5	JOHNS HOPKINS UNIVERSITY APPLIED PHYSICS LAB T R BETZER A R EATON R H KEITH D K PACE R L WEST JOHNS HOPKINS ROAD LAUREL MD 20723	1	UNIVERSITY OF CHICAGO DEPT OF THE GEOPHYSICAL SCIENCES G H MILLER 5734 S ELLIS AVE CHICAGO IL 60637
1	LOUISIANA STATE UNIVERSITY R W COURTER 948 WYLIE DR BATON ROUGE LA 70808	2	UNIVERSITY OF DAYTON RSCH INST N BRAR A PIEKUTOWSKI 300 COLLEGE PARK DAYTON OH 45469-0182
1	NC STATE UNIVERSITY Y HORIE RALEIGH NC 27695-7908	3	UNIVERSITY OF DELAWARE DEPT OF MECHANICAL ENGINEERING J GILLESPIE J VINSON D WILKINS NEWARK DE 19716
4	SOUTHWEST RESEARCH INSTITUTE C ANDERSON S A MULLIN J RIEGEL J WALKER PO DRAWER 28510 SAN ANTONIO TX 78228-0510	1	UNIVERSITY OF ILLINOIS PHYSICS BUILDING A V GRANATO URBANA IL 61801
1	SUNY STONEYBROOK DEPT APPL MATH & STAT J GLIMM STONEYBROOK NY 11794	1	UNIVERSITY OF MARYLAND PHYSICS DEPT BLDG 082 COLLEGE PARK MD 20742
1	UC BERKELEY MECHANICAL ENGINEERING DEPT GRADUATE OFFICE K LI BERKELEY CA 94720	1	UNIVERSITY OF PUERTO RICO DEPT CHEMICAL ENGINEERING L A ESTEVEZ MAYAGUEZ PR 00681-5000
2	UC SAN DIEGO DEPT APPL NECH & ENGR SVCS R011 S NEMAT-NASSER M MEYERS LA JOLLA CA 92093-0411	1	UNIVERSITY OF TEXAS DEPT OF MECHANICAL ENGINEERING E P FAHRENTHOLD AUSTIN TX 78712
2	UNIV OF ALA HUNTSVILLE AEROPHYSICS RSCH CTR G HOUGH D J LIQUORNIK PO BOX 999 HUNTSVILLE AL 35899	2	VIRGINIA POLYTECHNIC INSTITUTE COLLEGE OF ENGINEERING DEPT ENGNNG SCIENCE & MECHANICS R C BATRA BLACKSBURG VA 24061-0219
1	UNIV OF ALA HUNTSVILLE MECH ENGRNG DEPT W P SCHONBERG HUNTSVILLE AL 35899	2	AEROJET J CARLEONE S KEY PO BOX 13222 SACRAMENTO CA 95813-6000
		2	AEROJET ORDNANCE P WOLF G PADGETT 1100 BULLOCH BLVD SOCORRO NM 87801

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	ALLIANT TECHSYSTEMS INC R STRYK P SWENSON MN11-2720 600 SECOND ST NE HOPKINS MN 55343	3	DOW CHEMICAL INC ORDNANCE SYSTEMS C HANEY A HART B RAFANIELLO 800 BUILDING MIDLAND MI 48667
1	M L ALME 2180 LOMA LINDA DR LOS ALAMOS NM 87544-2769	3	DE TECHNOLOGIES INC P C CHOU R CICCARELLI W FLIS 3620 HORIZON DRIVE KING OF PRUSSIA PA 19406
1	APPLIED RESEARCH ASSOC INC J D YATTEAU 5941 S MIDDLEFIELD RD STE 100 LITTLETON CO 80123	3	DYNASEN J CHAREST M CHAREST M LILLY 20 ARNOLD PL GOLETA CA 93117
2	APPLIED RESEARCH ASSOC INC D GRADY F MAESTAS SUITE A220 4300 SAN MATEO BLVD NE ALBUQUERQUE NM 87110	1	ELORET INSTITUTE D W BOGDANOFF MS 230 2 NASA AMES RESEARCH CENTER MOFFETT FIELD CA 94035
1	APPLIED RESEARCH LABORATORIES T M KIEHNE PO BOX 8029 AUSTIN TX 78713-8029	1	EXPLOSIVE TECHNOLOGY M L KNAEBEL PO BOX KK FAIRFIELD CA 94533
1	ATA ASSOCIATES W ISBELL PO BOX 6570 SANTA BARBARA CA 93111	1	GB TECH LOCKHEED J LAUGHMAN 2200 SPACE PARK SUITE 400 HOUSTON TX 77258
1	BRIGS CO J E BACKOFEN 2668 PETERSBOROUGH ST HERNDON VA 20171-2443	2	GB TECH LOCKHEED L BORREGO C23C J FALCON JR C23C 2400 NASA ROAD 1 HOUSTON TX 77058
1	CENTURY DYNAMICS INC N BIRNBAUM 1001 GALAXY WAY SUITE 325 CONCORD CA 94520	6	GDLS 38500 MOUND RD W BURKE MZ436 21 24 G CAMPBELL MZ436 30 44 D DEBUSSCHER MZ436 20 29 J ERIDON MZ436 21 24 W HERMAN MZ 435 01 24 S PENTESCU MZ436 21 24 STERLING HTS MI 48310-3200
1	COMPUTATIONAL MECHANICS CONSULTANTS J A ZUKAS PO BOX 11314 BALTIMORE MD 21239-0314	1	GENERAL RESEARCH CORP T MENNA PO BOX 6770 SANTA BARBARA CA 93160-6770
1	CYPRESS INTERNATIONAL A CAPONECCHI 1201 E ABINGDON DR ALEXANDRIA VA 22314		
1	DESKIN RESEARCH GROUP INC E COLLINS 2270 AGNEW RD SANTA CLARA CA 95054		

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	RAYTHEON MSL SYS CO T STURGEON BLDG 805 MS D4 PO BOX 11337 TUCSON AZ 85734-1337	1	LIVERMORE SOFTWARE TECH CORP J O HALLQUIST 2876 WAVERLY WAY LIVERMORE CA 94550
5	INST FOR ADVANCED TECHNOLOGY S J BLESS J CAZAMIAS J DAVIS H D FAIR D LITTLEFIELD 3925 W BRAKER LN SUITE 400 AUSTIN TX 78759-5316	1	LOCKHEED MARTIN ELEC & MSLS G W BROOKS 5600 SAND LAKE RD MP 544 ORLANDO FL 32819-8907
1	INTERNATIONAL RESEARCH ASSOC D L ORPHAL 4450 BLACK AVE PLEASANTON CA 94566	1	LOCKHEED MARTIN MISSILE & SPACE W R EBERLE PO BOX 070017 HUNTSVILLE AL 35807
1	ITT SCIENCES AND SYSTEMS J WILBECK 600 BLVD SOUTH SUITE 208 HUNTSVILLE AL 35802	3	LOCKHEED MARTIN MISSILE & SPACE M A LEVIN ORG 81 06 BLDG 598 M R MCHENRY T A NGO ORG 81 10 BLDG 157 111 LOCKHEED WAY SUNNYVALE CA 94088
1	R JAMESON 624 ROWE DR ABERDEEN MD 21001	4	LOCKHEED MISSILE & SPACE CO J R ANDERSON W C KNUDSON S KUSUMI 0 81 11 BLDG 157 J PHILLIPS 0 54 50 PO BOX 3504 SUNNYVALE CA 94088
1	KAMAN SCIENCES CORP D L JONES 2560 HUNTINGTON AVE SUITE 200 ALEXANDRIA VA 22303	1	LOCKHEED MISSILE & SPACE CO R HOFFMAN SANTA CRUZ FACILITY EMPIRE GRADE RD SANTA CRUZ CA 95060
7	KAMAN SCIENCES CORP J ELDER R P HENDERSON D A PYLES F R SAVAGE J A SUMMERS T W MOORE T YEM 600 BLVD S SUITE 208 HUNTSVILLE AL 35802	1	MCDONNELL DOUGLAS ASTRONAUTICS CO B L COOPER 5301 BOLSA AVE HUNTINGTON BEACH CA 92647
3	KAMAN SCIENCES CORP S JONES G L PADEREWSKI R G PONZINI 1500 GRDN OF THE GODS RD COLORADO SPRINGS CO 80907	2	NETWORK COMPUTING SERVICES INC T HOLMQUIST G JOHNSON 1200 WASHINGTON AVE S MINNEAPOLIS MN 55415
1	D R KENNEDY & ASSOC INC D KENNEDY PO BOX 4003 MOUNTAIN VIEW CA 94040	3	GD OTS D A MATUSKA M GUNGER J OSBORN 4565A COMMERCIAL DR NICEVILLE FL 32578

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	PHYSICAL SCIENCES INC P NEBOLSINE 20 NEW ENGLAND BUS CTR ANDOVER MA 01810	2	TELEDYNE BROWN ENGR J W BOOTH M B RICHARDSON PO BOX 070007 MS 50 HUNTSVILLE AL 35807-7007
2	GD OTS D BOEKA N OUYE 400 ESTUDILLO AVE SUITE 100 SAN LEANDRO CA 94577-0205	1	ZERNOW TECHNICAL SVCS INC L ZERNOW 425 W BONITA AVE SUITE 208 SAN DIMAS CA 91773
1	PRC INC J ADAMS 5166 POTOMAC DR #103 KING GEORGE VA 22485-5824	44	<u>ABERDEEN PROVING GROUND</u> DIR USARL AMSRL WM BC A ZIELINSKI AMSRL WM MB G GAZONAS C HOPPEL AMSRL WM MC E CHIN J LASALVIA AMSRL WM T T HAVEL T W WRIGHT AMSRL WM TA W BRUCHEY W GILLICH M BURKINS W A GOOCH H W MEYER M NORMANDIA J RUNYEON AMSRL WM TB P BAKER R LOTTERO J STARKENBERG AMSRL WM TC R COATES T W BJERKE E KENNEDY K KIMSEY D SCHEFFLER S SCHRAML G SILSBY B SORENSEN R SUMMERS W WALTERS (3 CPS)
1	RAYTHEON ELECTRONIC SYSTEMS R LLOYD 50 APPLE HILL DRIVE TEWKSBURY MA 01876		
1	ROCKWELL INTERNATIONAL ROCKETDYNE DIVISION H LEIFER 16557 PARK LN CIRCLE LOS ANGELES CA 90049		
1	ROCKWELL MISSILE SYS DIV T NEUHART 1800 SATELLITE BLVD DULUTH GA 30136		
1	SAIC M W MCKAY 10260 CAMPUS POINT DR SAN DIEGO CA 92121		
1	SHOCK TRANSIENTS INC D DAVISON BOX 5357 HOPKINS MN 55343		
2	SOUTHERN RESEARCH INSTITUTE L A DECKARD D P SEGERS PO BOX 55305 BIRMINGHAM AL 35255-5305		
5	SRI INTERNATIONAL J D COLTON D CURRAN R KLOOP R L SEAMAN D A SHOCKEY 333 RAVENSWOOD AVE MENLO PARK CA 94025		

NO. OF
COPIES

ORGANIZATION

ABERDEEN PROVING GROUND (CONT)

AMSRL WM TD
E J RAPACKI
R BITTING
J COX
D DANDEKAR
K FRANK
M RAFTENBERG
G RANDERS PEHRSON (LLNL)
M SCHEIDLER
S SCHOENFELD
S SEGLETES (3 CPS)
T WEERISOORIYA
AMSRL WM TE
J POWELL
A PRAKASH

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	AERONAUTICAL & MARITIME RESEARCH LABORATORY S CIMPOERU D PAUL PO BOX 4331 MELBOURNE VIC 3001 AUSTRALIA	1	CEA R CHERET CEDEX 15 313 33 RUE DE LA FEDERATION PARIS 75752 FRANCE
1	DSTO AMRL WEAPONS SYSTEMS DIVISION N BURMAN RLLWS SALISBURY SOUTH AUSTRALIA 5108 AUSTRALIA	1	CEA CISI BRANCH P DAVID CENTRE DE SACLAY BP 28 GIF SUR YVETTE 91192 FRANCE
1	PRB S A M VANSNICK AVENUE DE TERVUEREN 168 BTE 7 BRUSSELS B 1150 BELGIUM	1	CEA CESTA A GEILLE BOX 2 LE BARP 33114 FRANCE
1	ROYAL MILITARY ACADEMY G DYCKMANS RENAISSANCELAAN 30 1000 BRUSSELS BELGIUM	5	CENTRE D'ETUDES DE GRAMAT C LOUPIAS P OUTREBON J CAGNOUX C GALLIC J TRANCHET GRAMAT 46500 FRANCE
1	BULGARIAN ACADEMY OF SCIENCES SPACE RESEARCH INSTITUTE V GOSPODINOV 1000 SOFIA PO BOX 799 BULGARIA	6	CENTRE DE RECHERCHES ET D'ETUDES D'ARCUEIL D BOUVART C COTTENNOT S JONNEAUX H ORSINI S SERROR F TARDIVAL 16 BIS AVENUE PRIEUR DE LA COTE D'OR F94114 ARCUEIL CODEX FRANCE
1	CANADIAN ARSENALS LTD P PELLETIER 5 MONTEE DES ARSENAUX VILLIE DE GRADEUR PQ J5Z2 CANADA	1	DAT ETBS CETAM C ALTMAYER ROUTE DE GUERRY BOURGES 18015 FRANCE
1	DEFENCE RSCH ESTAB SUFFIELD D MACKAY RALSTON ALBERTA T0J 2N0 RALSTON CANADA	1	ETBS DSTI P BARNIER ROUTE DE GUERAY BOITE POSTALE 712 18015 BOURGES CEDEX FRANCE
1	DEFENCE RSCH ESTAB SUFFIELD C WEICKERT BOX 4000 MEDICINE HAT ALBERTA T1A 8K6 CANADA	1	DEFENCE RSCH ESTAB VALCARTIER ARMAMENTS DIVISION R DELAGRAVE 2459 PIE X1 BLVD N PO BOX 8800 CORCELETTE QUEBEC GOA 1R0 CANADA

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	FRENCH GERMAN RESEARCH INST P-Y CHANTERET CEDEX 12 RUE DE L'INDUSTRIE BP 301 F68301 SAINT LOUIS FRANCE	3	FRAUNHOFER INSTITUT FUER KURZZEITDYNAMIK ERNST MACH INSTITUT H ROTHENHAEUSLER H SENF E STRASSBURGER KLINGELBERG 1 D79588 EFRINGEN KIRCHEN GERMANY
5	FRENCH GERMAN RESEARCH INST H-J ERNST F JAMET P LEHMANN K HOOG H F LEHR CEDEX 5 5 RUE DU GENERAL CASSAGNOU SAINT LOUIS 68301 FRANCE	3	FRENCH GERMAN RESEARCH INST G WEIHRAUCH R HUNKLER E WOLLMANN POSTFACH 1260 WEIL AM RHEIN D 79574 GERMANY
1	BATTELLE INGENIEUTECHNIK GMBH W FUCHE DUESSELDORFFER STR 9 ESCHBORN D 65760 GERMANY	2	IABG M BORRMANN H G DORSCH EINSTEINSTRASSE 20 D 8012 OTTOBRUN B MUENCHEN GERMANY
1	CONDAT J KIERMEIR MAXIMILIANSTR 28 8069 SCHEYERN FERNHAG GERMANY	1	INGENIEURBUERO DEISENROTH AUF DE HARDT 33 35 D5204 LOHMAR 1 GERMANY
1	TDW M HELD POSTFACH 13 40 D 86523 SCHROBENHAUSEN GERMANY	1	TU MUENCHEN E IGENBERGS ARCISSTRASSE 21 8000 MUENCHEN 2 GERMANY
1	DIEHL GBMH AND CO M SCHILDKNECHT FISCHBACHSTRASSE 16 D 90552 ROETBENBACH AD PEGNITZ GERMANY	1	NATIONAL GEOPHYSICAL RESEARCH INSTITUTE G PARTHASARATHY HYDERABAD 500 007 AP INDIA
4	ERNST MACH INSTITUT V HOHLER E SCHMOLINSKE E SCHNEIDER K THOMA ECKERSTRASSE 4 D 7800 FREIBURG I BR 791 4 GERMANY	5	RAFAEL BALLISTICS CENTER E DEKEL Y PARTOM G ROSENBERG Z ROSENBERG Y YESHURUN PO BOX 2250 HAIFA 31021 ISRAEL
		1	TECHNION INST OF TECH FACULTY OF MECH ENGNG S BODNER TECHNION CITY HAIFA 32000 ISRAEL

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	IHI RESEARCH INSTITUTE STRUCTURE & STRENGTH T SHIBUE 1 15 TOYOSU 3 KOTO TOKYO 135 JAPAN	3	INST OF MECH ENGNRG PROBLEMS V BULATOV D INDEITSEV Y MESCHERYAKOV BOLSHOY 61 VO ST PETERSBURG 199178 RUSSIAN REPUBLIC
1	ESTEC CS D CASWELL BOX 200 NOORDWIJK 2200 AG NETHERLANDS	1	INSTITUTE OF MINEROLOGY & PETROGRAPHY V A DREBUSHCHAK UNIVERSITETSKI PROSPEKT 3 630090 NOVOSIBIRSK RUSSIAN REPUBLIC
2	EUROPEAN SPACE AGENCY ESTEC L BERTHOUD M LAMBERT POSTBUS BOX 299 NOORDWIJK NL2200 AG NETHERLANDS	2	IOFFE PHYSICO TECHNICAL INSTITUTE DENSE PLASMA DYNAMICS LABORATORY E M DROBYSHEVSKI A KOZHUSHKO ST PETERSBURG 194021 RUSSIAN REPUBLIC
4	PRINS MAURITS LABORATORY H J REITSMA E VAN RIET H PASMAN R YSSELSTEIN TNO BOX 45 RIJSWIJK 2280AA NETHERLANDS	1	IPE RAS A A BOGOMAZ DVORTSOVAIA NAB 18 ST PETERSBURG RUSSIAN REPUBLIC
1	ROYAL NETHERLANDS ARMY J HOENEVELD V D BURCHLAAN 31 PO BOX 90822 2509 LS THE HAGUE NETHERLANDS	2	LAVRENTYEV INST HYDRODYNAMICS L A MERZHIEVSKY V V SILVESTROV 630090 NOVOSIBIRSK RUSSIAN REPUBLIC
1	INSTITUTE OF CHEMICAL PHYSICS A YU DOLGOBORODOV KOSYGIN ST 4 V 334 MOSCOW RUSSIAN REPUBLIC	1	MOSCOW INST OF PHYSICS & TECH S V UTYUZHNIKOV DEPT OF COMPUTATIONAL MATHEMATICS DOLGOPRUDNY 1471700 RUSSIAN REPUBLIC
4	INSTITUTE OF CHEMICAL PHYSICS RUSSIAN ACADEMY OF SCIENCES G I KANEL A M MOLODETS S V RAZORENOV A V UTKIN 142432 CHERNOGOLOVKA MOSCOW REGION RUSSIAN REPUBLIC	1	RESEARCH INSTITUTE OF MECHANICS NIZHNIY NOVGOROD STATE UNIVERSITY A SADYRIN P R GAYARINA 23 KORP 6 NIZHNIY NOVGOROD 603600 RUSSIAN REPUBLIC
		2	RUSSIAN FEDERAL NUCLEAR CENTER VNIIEF L F GUDARENKO R F TRUNIN MIRA AVE 37 SAROV 607190 RUSSIAN REPUBLIC

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	SAMARA STATE AEROSPACE UNIV L G LUKASHEV SAMARA RUSSIAN REPUBLIC	2	K&W THUN W LANZ W ODERMATT ALLMENDSSTRASSE 86 CH-3602 THUN SWITZERLAND
1	TOMSK BRANCH OF THE INSTITUTE FOR STRUCTURAL MACROKINETICS V GORELSKI 8 LENIN SQ GSP 18 TOMSK 634050 RUSSIAN REPUBLIC	2	AWE M GERMAN W HARRISON FOULNESS ESSEX SS3 9XE UNITED KINGDOM
1	UNIVERSIDAD DE CANTABRIA FACULTAD DE CIENCIAS DEPARTAMENTO DE FISICA APLICADA J AMOROS AVDA DE LOS CASTROS SN 39005 SANTANDER SPAIN	1	CENTURY DYNAMICS LTD N FRANCIS DYNAMICS HOUSE HURST RD HORSHAM WEST SUSSEX RH12 2DT UNITED KINGDOM
1	CARLOS III UNIV OF MADRID C NAVARRO ESCUELA POLITEENICA SUPERIOR C BUTARQUE 15 28911 LEGANES MADRID SPAIN	5	DERA I CULLIS J P CURTIS Q13 A HART Q13 K COWAN Q13 M FIRTH R31 FORT HALSTEAD SEVENOAKS KENT TN14 7BP UNITED KINGDOM
1	DYNAMEC RESEARCH AB A PERSSON PO BOX 201 S 151 23 SODERTALJE SWEDEN	6	DEFENCE RESEARCH AGENCY W A J CARSON I CROUCH C FREW T HAWKINS B JAMES B SHRUBSALL CHOBHAM LANE CHERTSEY SURREY KT16 0EE UNITED KINGDOM
7	FOI SWEDISH DEFENCE RESEARCH AGENCY GRINDSJON RESEARCH CENTRE L GUNNAR OLSSON B JANZON G WIJK R HOLMLIN C LAMNEVIK L FAST M JACOB SE 147 25 TUMBA SWEDEN	1	UK MINISTRY OF DEFENCE G J CAMBRAY CBDE PORTON DOWN SALISBURY WITTSIRE SPR 0JQ UNITED KINGDOM
2	SWEDISH DEFENCE RSCH ESTAB DIVISION OF MATERIALS S J SAVAGE J ERIKSON STOCKHOLM S17290 SWEDEN		

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
7	INSTITUTE FOR PROBLEMS IN MATERIALS SCIENCE S FIRSTOV B GALANOV O GRIGORIEV V KARTUZOV V KOVTUN Y MILMAN V TREFILOV 3 KRHYZHANOVSKY STR 252142 KIEV 142 UKRAINE
1	INSTITUTE FOR PROBLEMS OF STRENGTH G STEPANOV TIMIRYAZEVSKEYU STR 2 252014 KIEV UKRAINE