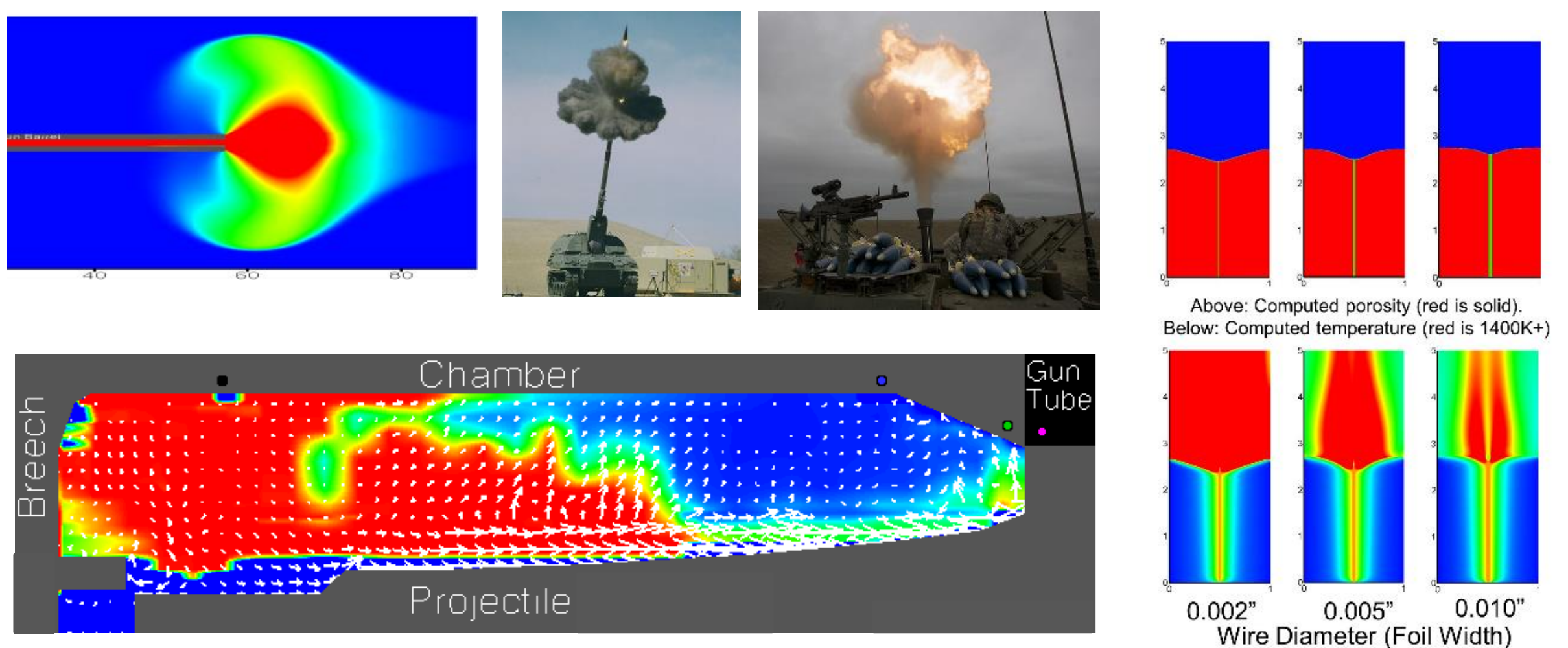


S&T Campaign: Sciences for Lethality and Protection
Kinetic Lethality
Propulsion and Launch

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Research Objective

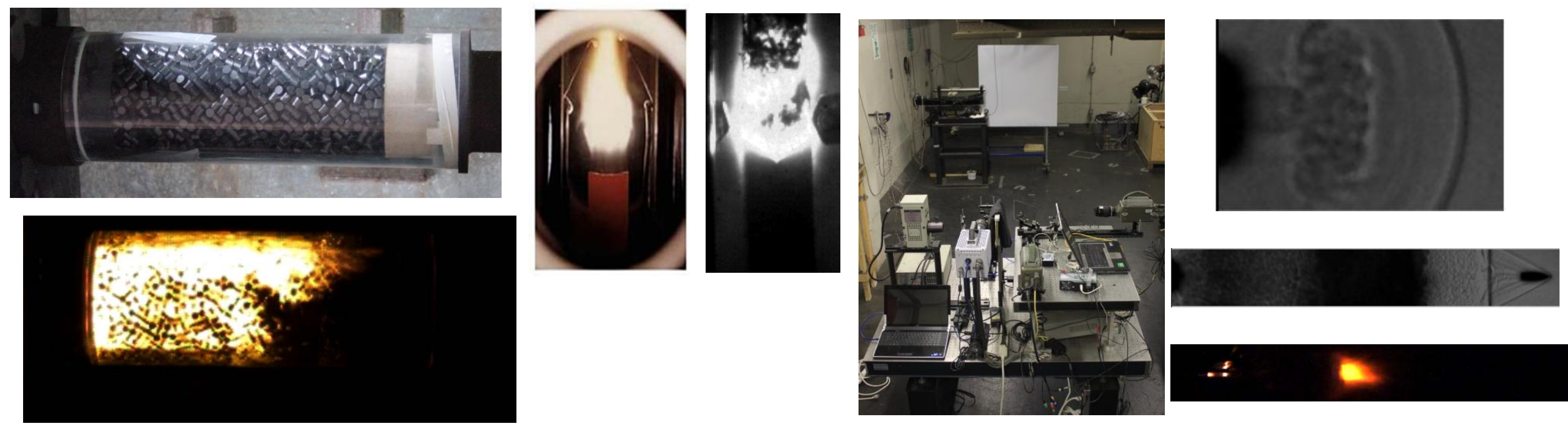
- Understanding the physics of gun interior ballistics, gun muzzle blast/flash, and rocket internal ballistics.
- Develop and validate high-fidelity modeling capabilities to predict multidimensional, multiphase, chemically-reacting flow fields for solid propellant gun charges and rocket engines used for tactical armament propulsion.
- Develop and validate chemical kinetics mechanisms for solid propellants used in Army guns as well as minimum-smoke and composite propellants used in Army missiles.
- Create synergism between computational physics-based models and experimental facilities (full- and laboratory-scale).



Gun muzzle blast/flash with model; gun charge modeling; strand model.

ARL Facilities and Capabilities Available to Support Collaborative Research

- Expertise in physics-based computational model development (lumped-parameter, CFD, burnrate models) for gun charges and tactical rocket engines.
- Expertise in gas-phase chemical kinetics mechanism generation/customization for solid propellants developed for use in gun charges, igniters, and rockets.
- Complementary computational and experimental resources for gun interior ballistics, gun muzzle blast, and solid propellant strand burn visualization and rate data.
- Gun ballistic ranges and laboratory-scale test facilities.
- DOD Supercomputing Resource Center (DSRC).



Select experimental facilities (CW from upper left: windowed strand burner, muzzle blast/flash Schlieren, gun interior ballistics simulator).

Challenges

- Mitigating high-pressure fields associated with gun muzzle blast (pressure, temperature, signature).
- Fundamental understanding of the interaction of reaction kinetics and high-pressure, multi-phase fluid dynamics in propulsion systems (guns, rocket engines).
- Reducing kinetics mechanisms to computationally tractable sizes while maintaining accuracy over wide range of pressures, temperatures, and compositions.
- Difficult obtaining quality validation data over range of conditions for strand-burners and small rocket engines.
- Developing efficient computational methods and algorithms.

Complementary Expertise / Facilities / Capabilities Sought in Collaboration

- Mitigation techniques for high-pressure fields produced by blast from propulsion systems (e.g., gun muzzle blast).
- Precision bench-level experiments for the measurement of gun charge dynamics during ignition-phase.
- High-quality validation data (propellant strand burnrate and chemical composition, engine internal pressure and temperature) over wide range of operating conditions (10,000+ psi) and propellant compositions.
- Shock tube data to support gas-phase chemical kinetics mechanism development for specific critical reactions.
- Solid (condensed) phase kinetics for high-pressure, multi-phase, chemically-reacting flows.

Reaction	Rate Coefficient	Order	Ea
1. NO2+H2O=HNO+OH	2.45E12	0.0	4271
2. NO2+NO2=HNO+NO3	8.46E09	0.23	20252
3. NO2+NO2=HNO+NO+O2	6.51E12	0.0	27609
4. NO2+NO2=HNO+NO2+O2	2.11E12	0.0	25000
5. NO2+NO2=NO2+HNO	1.30E13	0.0	28590
6. NO2+NO2=HNO2+NO	1.20E13	0.0	25000
7. NO2+NO2=HNO2+NO+O	4.42E04	2.44	4042
8. NO2+NO2=HNO2+NO2	1.20E13	0.0	5943
9. NO2+OH=HNO2+NO2	1.27E10	1.0	135
10. NO2+OH=HNO2+NO	8.20E09	1.34	130

Gas-phase chemical kinetics mechanism research & development.